



Starshade Exoplanet Mission Options WFIRS Conference

Doug Lisman

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Starshade Spacecraft Architecture



Exoplanet Exploration Program

Compact stowed volume fits in 5m fairing

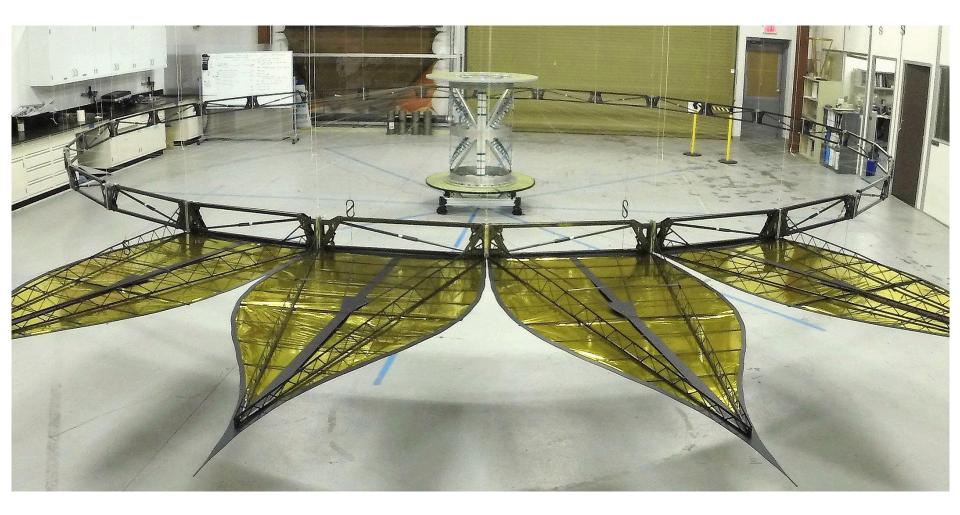
Inner Disk formed by a perimeter truss, wire spokes, optical shield and central hub Another spacecraft (e.g., telescope) can stack on top

Bus system mounts to central hub with optional propellant tanks in center

Petals formed by a furlable lattice structure, optical shields and pop-up ribs Optical edges define
apodization function and control solar glint







10m diameter inner disk and 3.5m long petals assembled by undergraduate students in summer of 2014





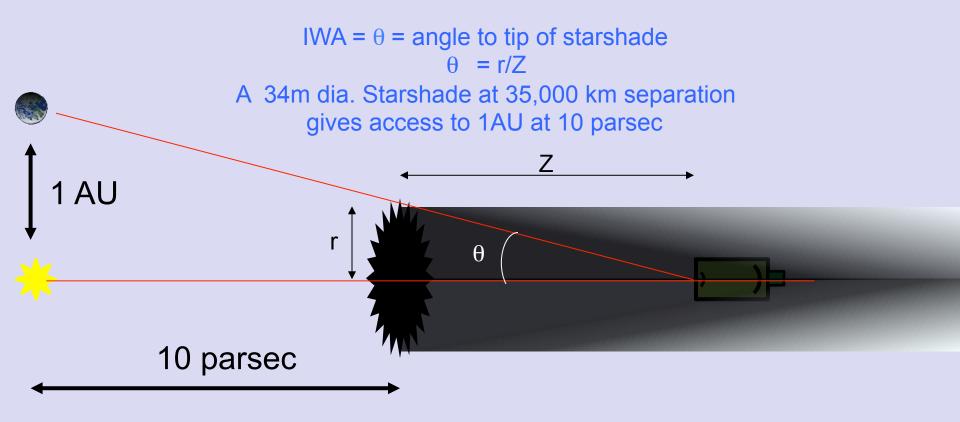
Deployment tolerances were successfully demonstrated with this earlier prototype with 20 deployment cycles

Studying Other Worlds with the Help of a Starshade





- Starlight is suppressed before entering the telescope, so wavefront does not require correction and no stringent requirements are levied on the telescope
- No outer working angle
- Contrast and inner working angle are largely independent of the telescope
- Lateral formation control band is created by making the shadow larger than aperture





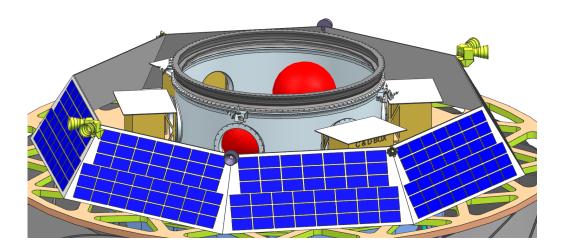


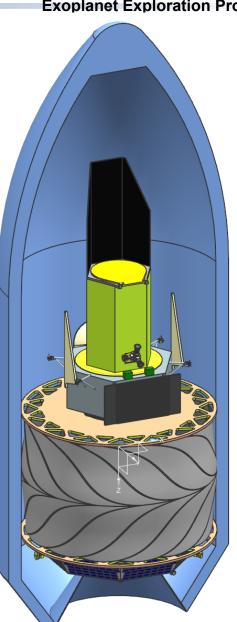
- Exo-S STDT is studying two mission scenarios:
 - Small dedicated telescope co-launched with starshade
 - Larger multi-purpose telescope launched separate from starshade
- Remainder of presentation details the two options the STDT plans to detail in their report





- 1.1 m telescope stacks on-top starshade at launch •
- Falcon-9 launch vehicle with 5 m fairing
- Earth Drift Away orbit for simple navigation and benign environment for formation control
- Telescope provides propulsion for retargeting and • formation control
- Telescope carries instrumentation for starshade:
 - Field camera, guide camera and IFS with 3" FOV
- 30 m starshade operates 34,000 km from telescope for primary band of 515-825 nm

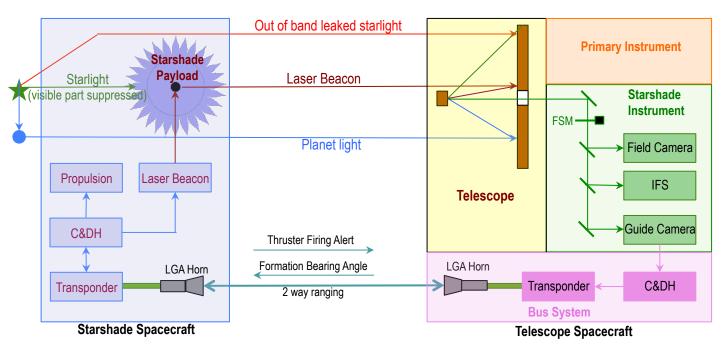








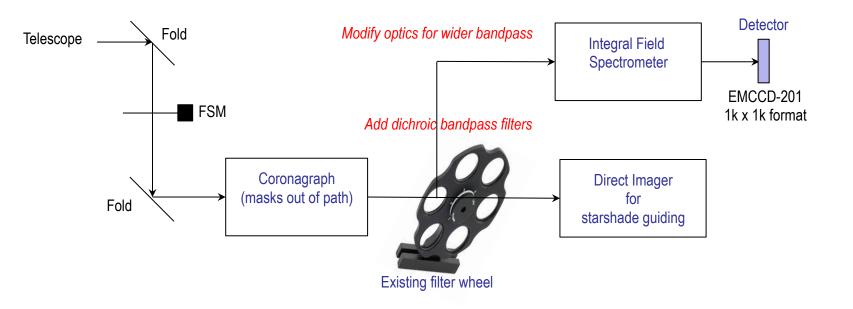
- Starshade based exoplanet science is a secondary mission objective
- Telescope launches with a "starshade ready package":
 - Proximity radio for inter-spacecraft communication and range measurement
 - Guide camera for sensing starshade lateral position
 - IFS for science and optional field camera
- Telescope operates at Earth-Sun L2 to enable rendezvous
- Starshade launches later to start observing after primary mission objectives are met
- Starshade provides propulsion for retargeting and formation control







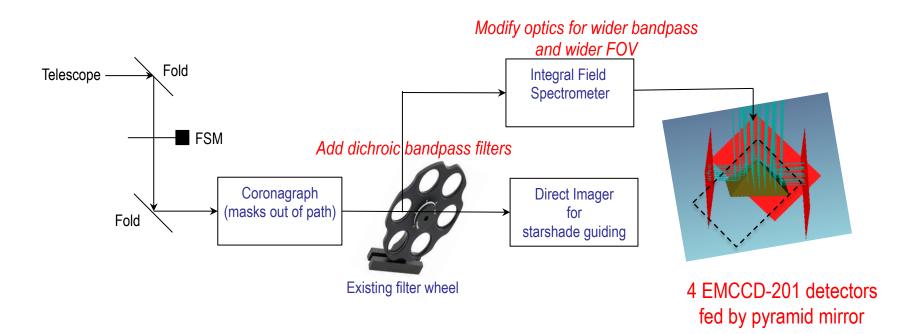
- Study assumes that WFIRST/AFTA operates at Earth-Sun L2
- Spacecraft bus provides proximity communications (transponder and LGA)
- Coronagraph provides starshade instrumentation with no added focal planes
 - Direct Imager for starshade guiding, IFS for science and no field camera (optional)
 - Dichroic bandpass filters added and IFS optics modified for wider bandpass
- Starshade FOV is limited to 0.75" by IFS detector size and coronagraph optimized pixel scale and spacing between spectrums
- 34 m starshade operates 35 Mm from telescope for primary band of 600-850 nm







- FOV increase to 1.1" requires adding detector area
- Low noise detector required for coronagraph is only available in 1k x 1k format
- This concept spreads spectrums over 4 baseline detectors fed by a pyramid mirror







- Falcon-9 launch vehicle with 5m fairing
- Simple spin-stabilized spacecraft (1/3 rpm), with loose pointing requirements (1 deg)
- Chemical propulsion with commercial tanks sized to fit in existing 1.6 m dia. Hub
- Propellant capacity is sufficient for > 3 years of operation, with large mass margins
 - Excess capacity could be used to deliver a separate payload to L2
- Future studies will consider SEP with solar cells embedded in optical shield

